

511/10/105

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METALS AND THERMAL STRUCTURES BRANCH
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SPACE TRANSPORTATION TECHNOLOGY WORKSHOP
MARSHALL SPACE FLIGHT CENTER
OCTOBER 11-12, 2000

Airframe/TPS

AIRFRAME DESIGN AND INTEGRATION

- ♦ **Integrated Design Tools and Methods**
- ♦ **Integrated Airframe Trade Studies**
NRA 8-21 Overview
- ♦ **Risk and Reliability Assessment**
LaRC Points of Contact:

Jeff Stroud

Analytical and Computational Methods Branch, Structures and Materials
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757-864-2928

Tom Zang

Multi-Disciplinary Design and Optimization Branch
t.a.zang@larc.nasa.gov
757-864-2307

Airframe/TPS

AIRFRAME DESIGN AND INTEGRATION: Major Areas

- ◆ Decompose operational, safety, and cost requirements into a comprehensive and consistent set of design criteria for different structural and material concepts for Reusable Launch Vehicles (RLVs)
- ◆ Develop compliance methods to ensure that different structural and material concepts are assessed at a consistent and adequate level of fidelity and safety
- ◆ Develop and assess weight reduction potential of integrated airframe concepts for RLVs, e.g. Thermal Protection System (TPS)/ TPS Support (TPSS)/ Cryogenic Tank (CT) System
- ◆ Compare performance and weight of various airframe structural and material concepts and structural arrangements and identify technology development needs
- ◆ Develop high fidelity parametric models that include airframe structural interactions and major design drivers

AIRFRAME INTEGRATION TRADE ^{Airframe/TPS}STUDIES

GENERAL OBJECTIVES

- ♦ Define vehicle requirements, definition, packaging
- ♦ Define airframe structural design requirements and develop compliance methods
- ♦ Define load conditions, loads, factors of safety, and materials
- ♦ Define integrated concepts
- ♦ Develop methods, perform analysis, and sizing
- ♦ Calculate system weights
- ♦ Assess concepts

Airframe/TPS

TRADE STUDIES: GENERAL APPROACH

RLV Requirements

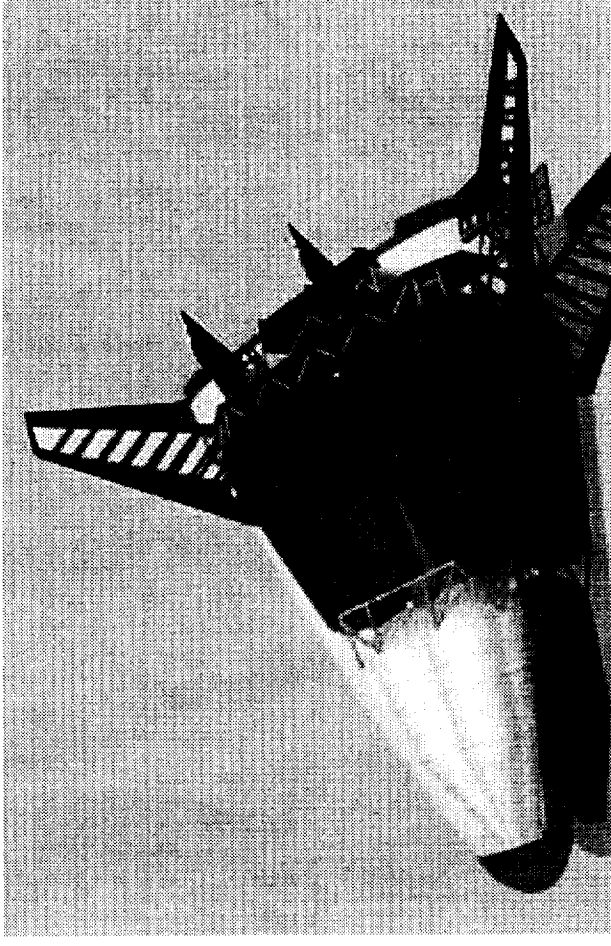
- Lightweight
- Fully reusable
- Easily maintained

Vehicle Definition

- Single Stage to Orbit (SSTO)
- Lifting body

Major Components

- Aerospike engines
- Engine Thrust Structure
(integrates engines, fins, tanks, and main landing gear)



- Liquid Oxygen (LOX) Tank
- Liquid Hydrogen (LH2) Tanks
- Intertank
- Metallic Thermal Protection System and support structure

Airframe/TPS

SYSTEM DEFINITION

♦ John T. Dorsey	Airframe Integration & Concepts, Vehicle Loads, Weights (Study Lead, NRA 8-21)
♦ Max Blosser	TPS Concepts (TPS Team Lead)
♦ Carl Poteet	TPS Thermal Analysis and Sizing, TPS Concepts
♦ Roger Chen	TPS Panel Acoustic, Fatigue, Creep Analysis & Sizing
♦ Irv Schmidt	TPS Panel and Tank Stiffening Design
♦ John Wang	Semi-Conformal & Lobed Tank Analysis and Sizing
♦ Su-Yuen Hsu	TPS Panel Structural Analysis and Sizing
♦ Lynn Bowman	Non-Optimum and Vehicle Weights
♦ Jeff Cerro	Material Properties, Vehicle Loads
♦ Nani Balakrishnan	TPS Panel Structural Analysis and Sizing
♦ David Myers	Packaging, Weights, & TPS
♦ Kevin Rivers	2 nd Gen RLV Airframe Integration and Trades Lead
♦ Kim Bey	Aerothermal Loads, Thermal Analysis, TPS Sizing

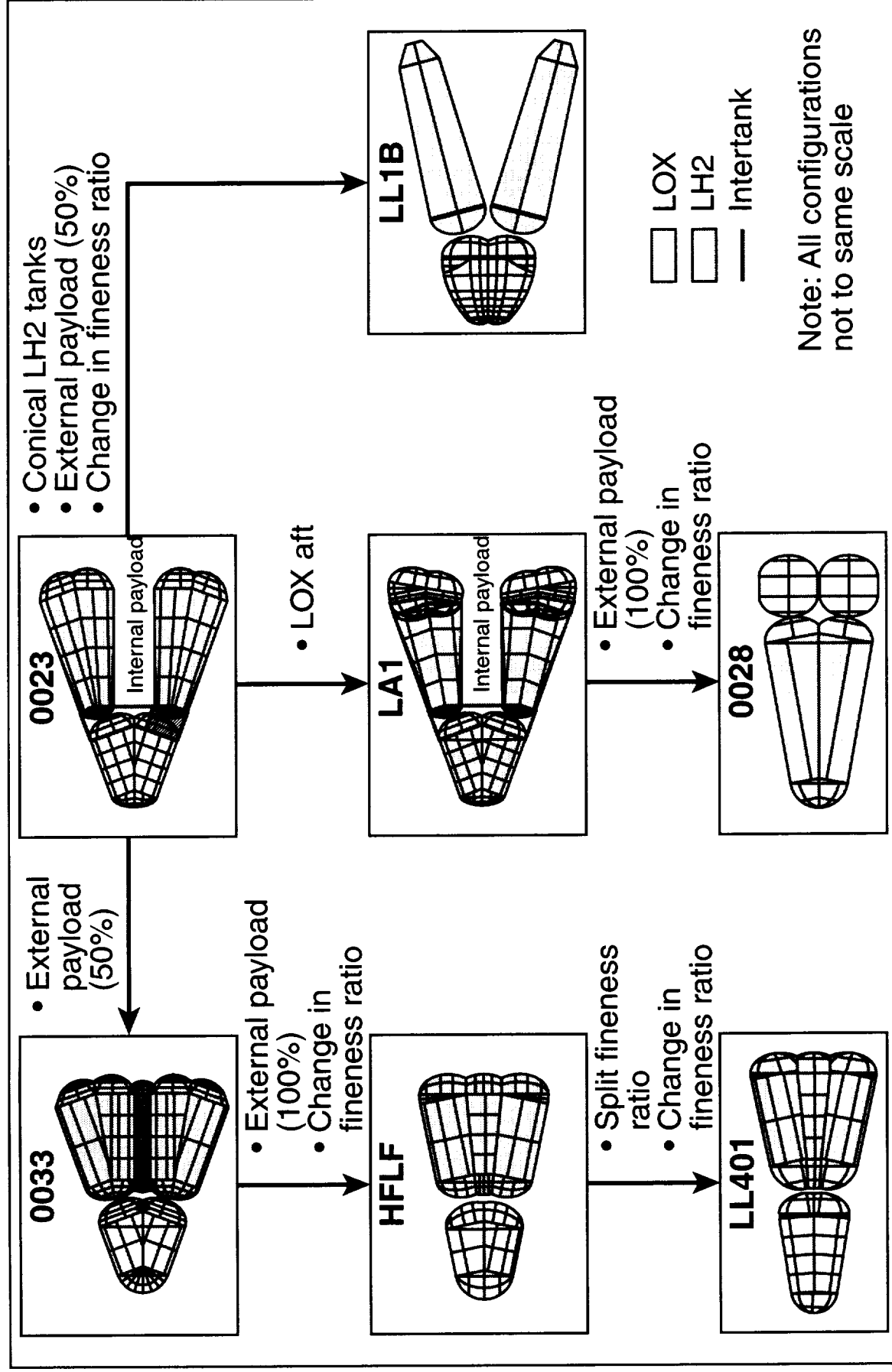
Airframe/TPS

TRADE STUDY TEAM

- ♦ Tank Packaging and Geometry
 - Packaging configurations
 - Lobed versus conformal tanks
 - Minimize Distance between Outer Mold Line (OML) and Tank
- ♦ Component Trade Studies
 - Tanks
 - TPS and support structure
- ♦ Integrated TPS, TPS support, and tanks
 - Applicable to several architectures
 - In progress

Airframe/TPS

TRADE STUDIES FOR NRA 8-21: MAJOR CATEGORIES



Airframe/TPS

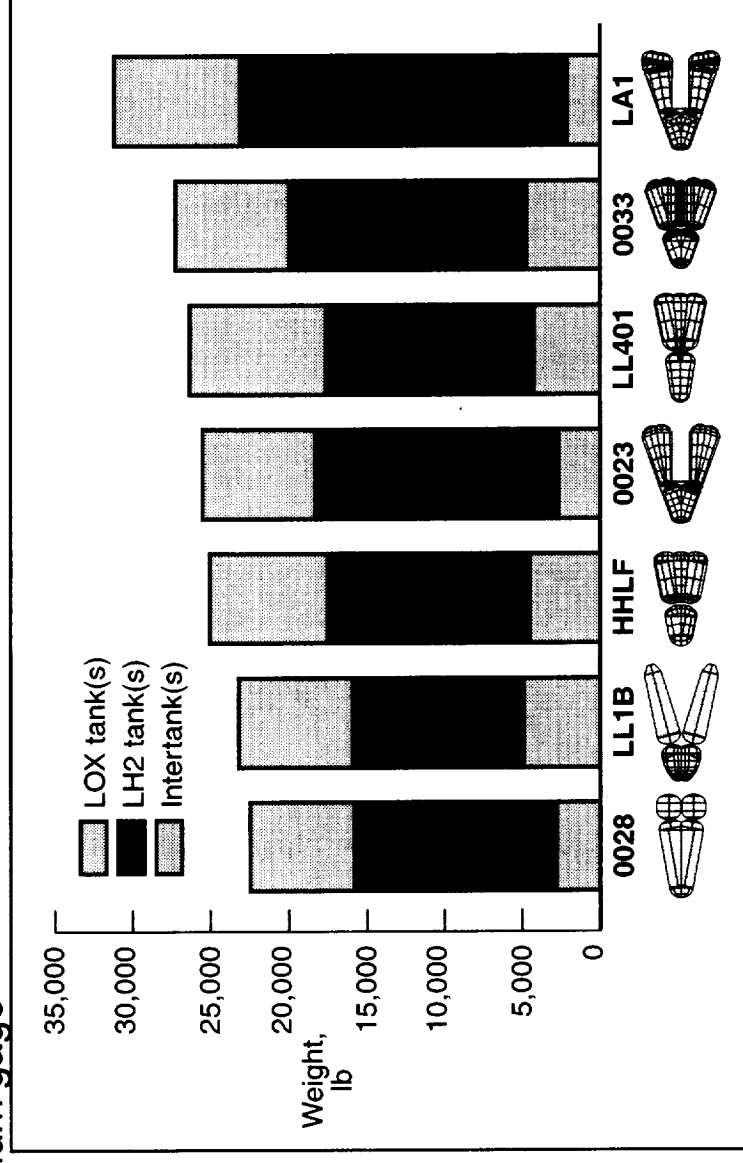
PACKAGING CONFIGURATIONS

• Material Specifications

- Quasi-isotropic PMC laminates
- Limit strain 6000 $\mu\text{in./in.}$
- Minimum gage

• Load Cases

- Launch (1.355 g's)
- Max acceleration (3 g's)



• Interactions:

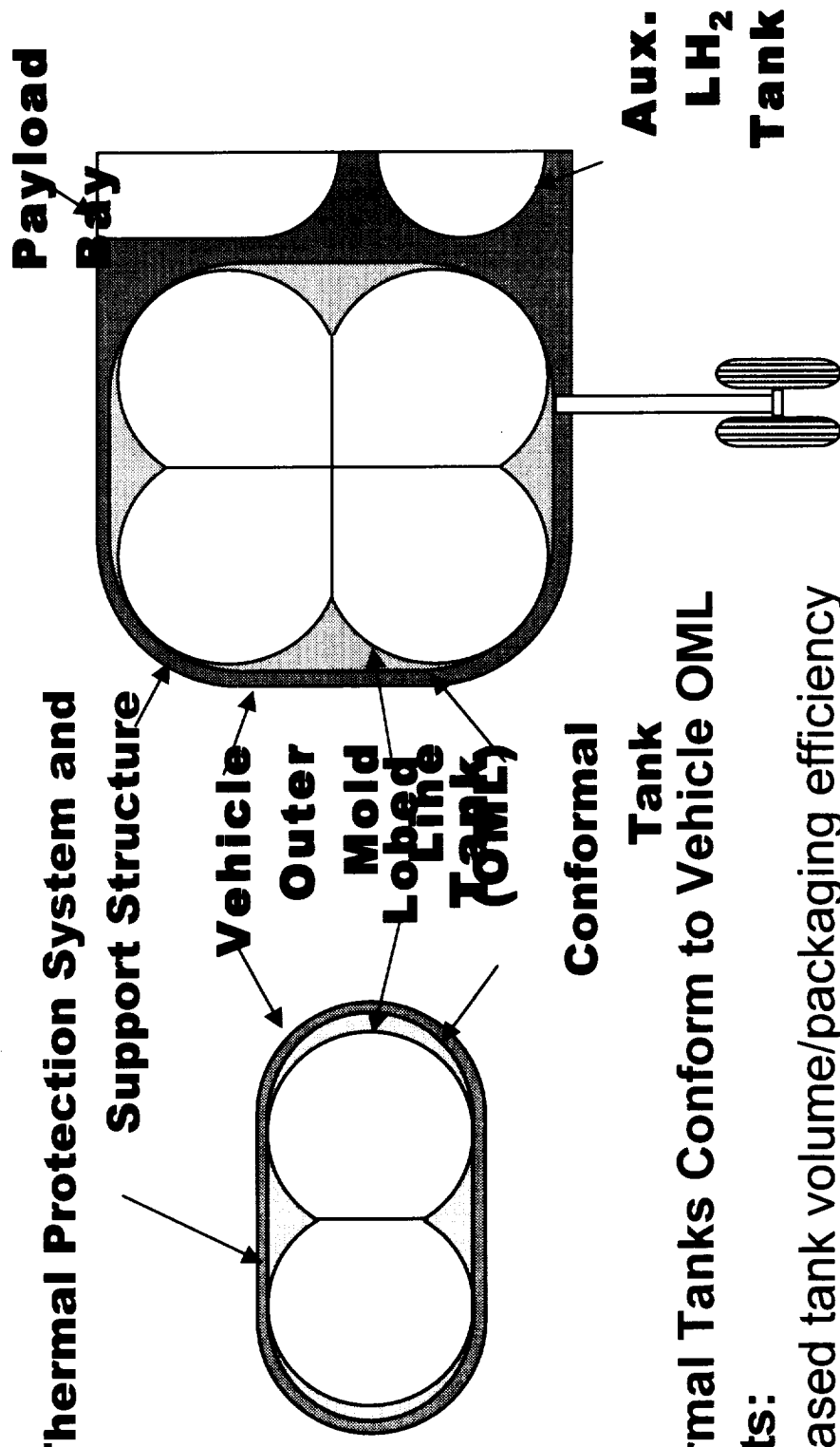
- LOX-aft: reduced LH2 ullage pressure for tank stabilization changes engine operating requirements
- External payload and aerodynamics

Airframe/TPS

CONFIGURATION SIZING

LOX Tank Geometry

LH2 Tank Geometry



Conformal Tanks Conform to Vehicle OML

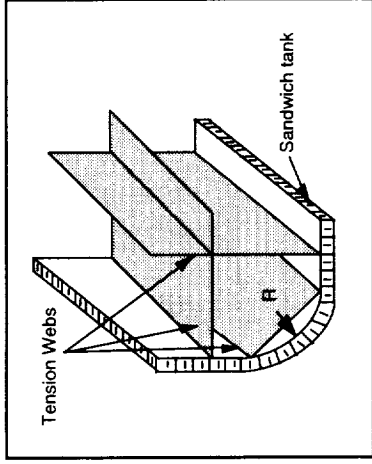
Benefits:

- ◆ Increased tank volume/packaging efficiency
- ◆ Reduced TPS support structure
- ◆ Improved thrust load paths

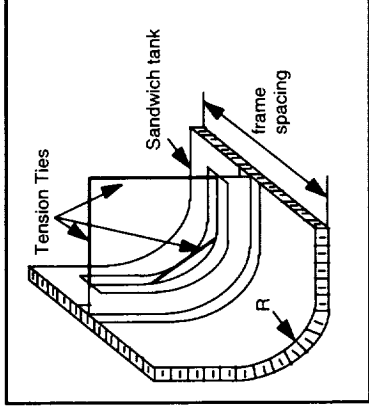
Airframe/TPS

TANK GEOMETRY TRADE STUDIES

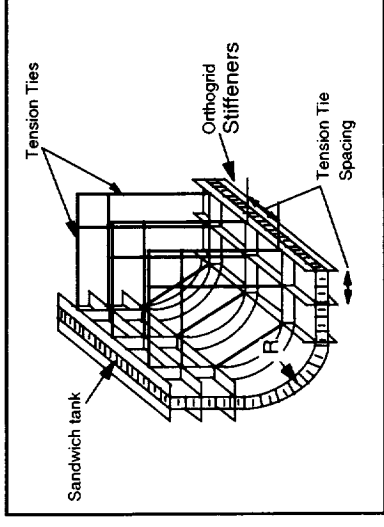
Axial Tension Webs



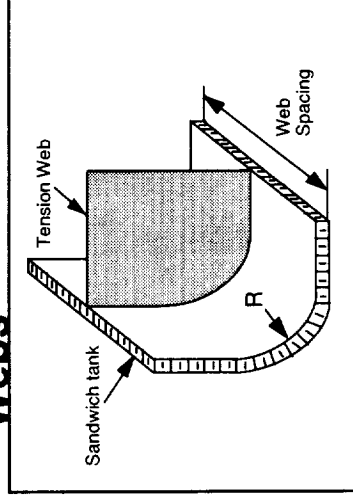
Frames & Ties



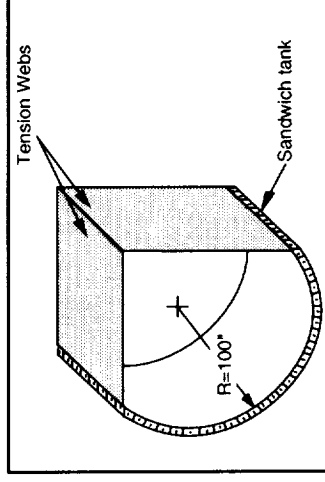
Orthogrid & Ties



Transverse Webs



Quad-lobe & Webs

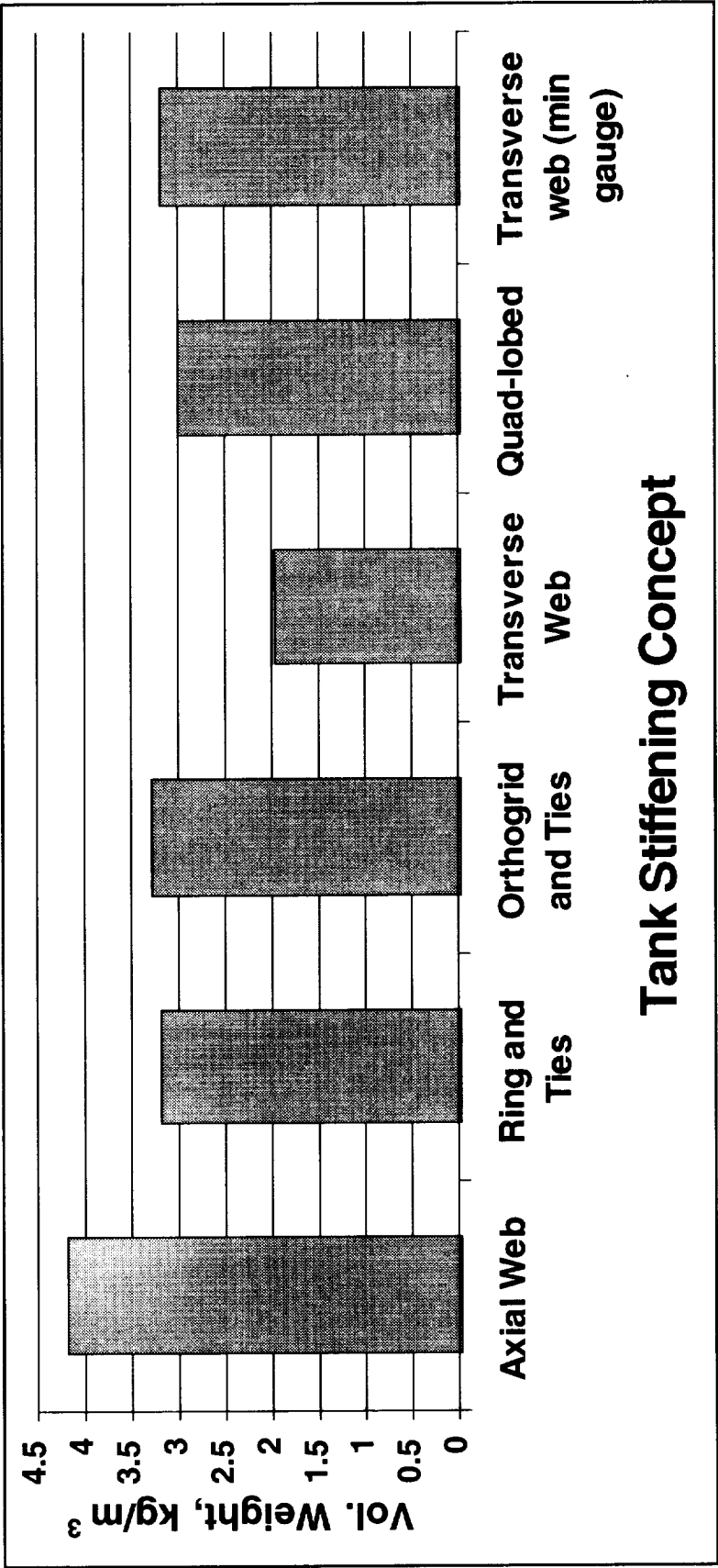


Airframe/TPS

TANK STIFFENING CONCEPTS CONSIDERED

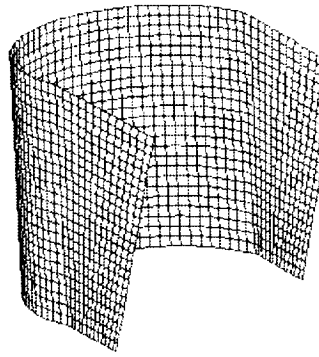
Corner radius = 0.635 m
Pressure = 137.9 KPa

No manufacturing min. gauge

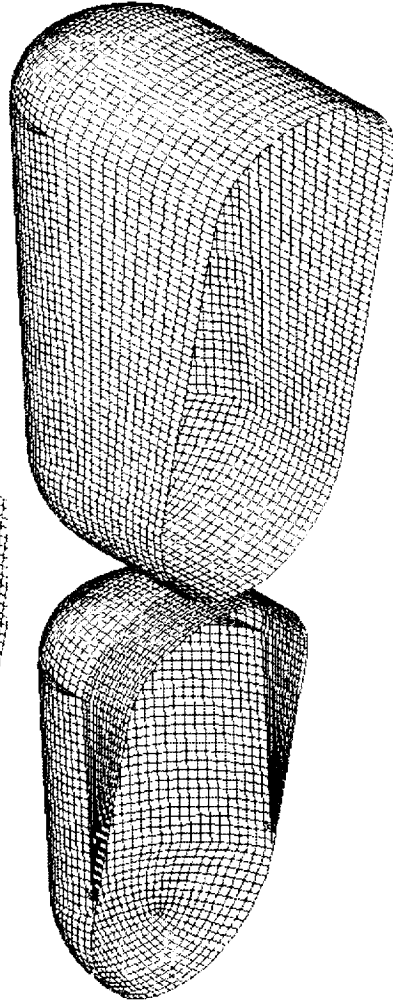


Airframe/TPS

TANK STIFFENING WEIGHTS COMPARISON

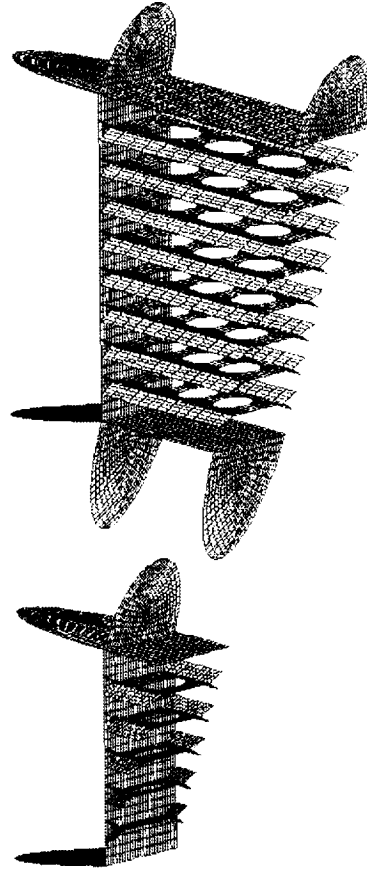


Intertank
Structure



LOX Tank

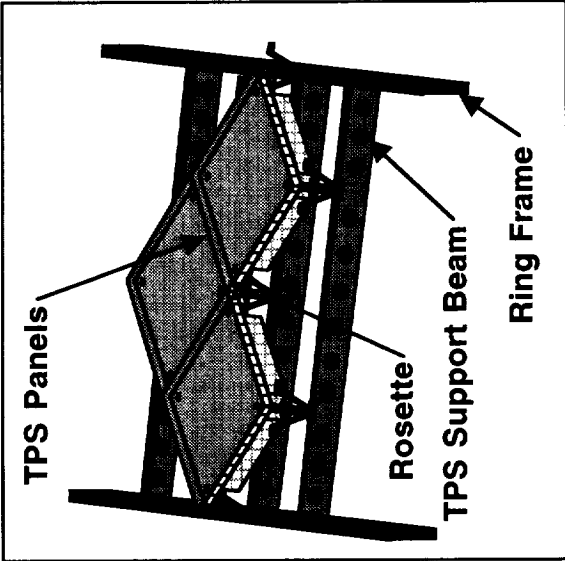
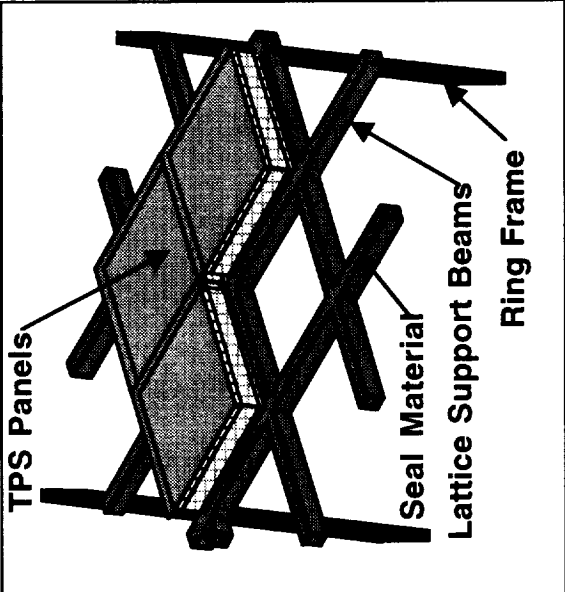
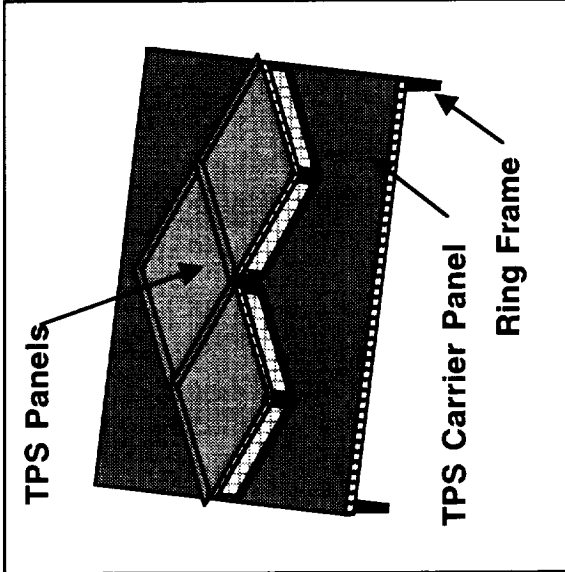
LH2 Tank



Tension Ties

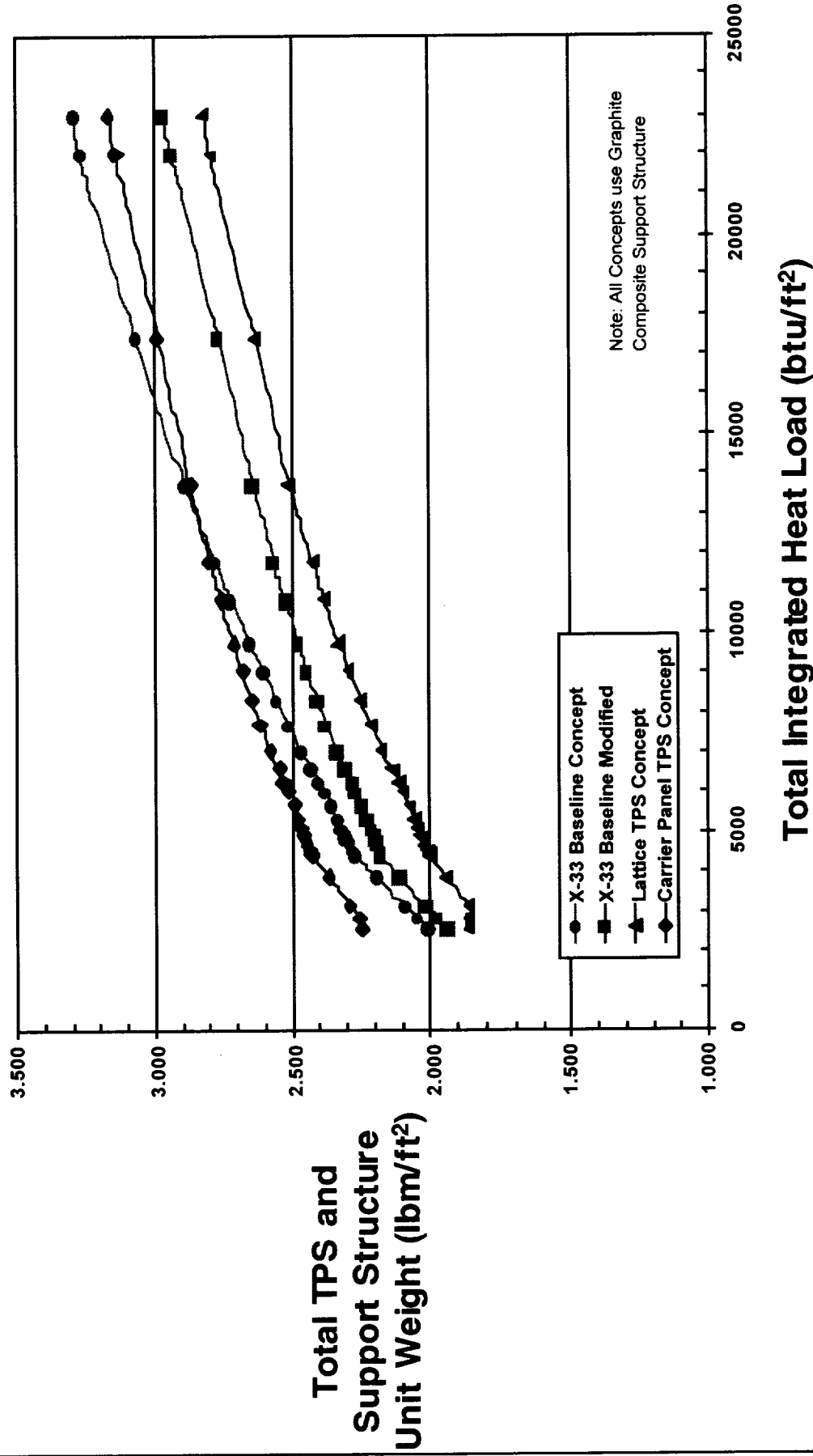
Airframe/TPS
SEMI-CONFORMAL SANDWICH TANK SYSTEM: FEM FOR SIZING
(In Progress)

Metallic TPS Concepts for Windward Aeroshell Surfaces

<p>Rohr X-33 Concept (baseline)</p>	 <p>TPS Panels</p> <p>Rosette</p> <p>TPS Support Beam</p> <p>Ring Frame</p>	<ul style="list-style-type: none"> •Thermal stresses minimized •Light weight system •Simplified manufacturing •Seals on hot surface •Hot surfaces carry aero pressures •Panel damage/loss potentially catastrophic
<p>LaRC-Type TPS Panels with Lattice Seal & Support Frames</p>	 <p>TPS Panels</p> <p>Seal Material</p> <p>Lattice Support Beams</p> <p>Ring Frame</p>	<ul style="list-style-type: none"> •Seals and pressure bearing surface moved to cooler region •Tolerant to outer surface damage •More complex support structure •More costly TPS panel •Panel loss potentially catastrophic
<p>LaRC-Type TPS Panels Mounted to Carrier Plates</p>	 <p>TPS Panels</p> <p>TPS Carrier Panel</p> <p>Ring Frame</p>	<ul style="list-style-type: none"> •Carrier panel increases TPS options •Improved damage tolerance •Reduces number of seals •Pressure carried on large, cool panel •Heavier than other concepts •Complicates removal for inspection •Potential thermal stress issues

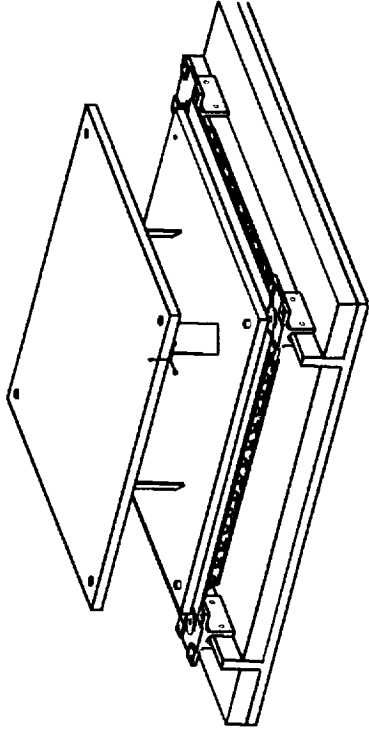
Airframe/TPS

TPS TRADE STUDY CONCEPTS



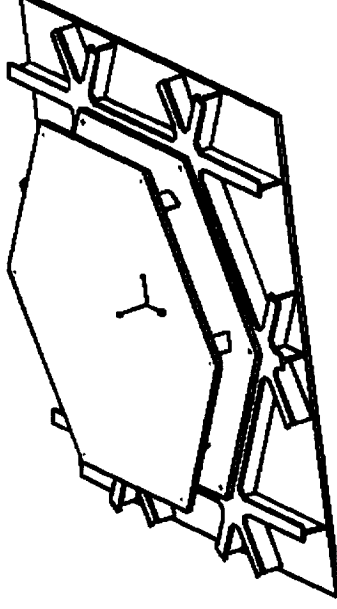
TPS AND SUPPORT STRUCTURE WEIGHTS

REPRESENTATIVE STRUCTURAL CONCEPTS



FEATURES

- Metallic TPS
- TPS Support Structure
- Purge Gap
- Cryogenic Foam
- Sandwich or Stiffened Skin Tank Wall



FEATURES

- Metallic TPS
- Direct Attach TPS
- Cryogenic Foam
- Sandwich, Stiffened Skin, or Integrally-Stiffened Tank Wall

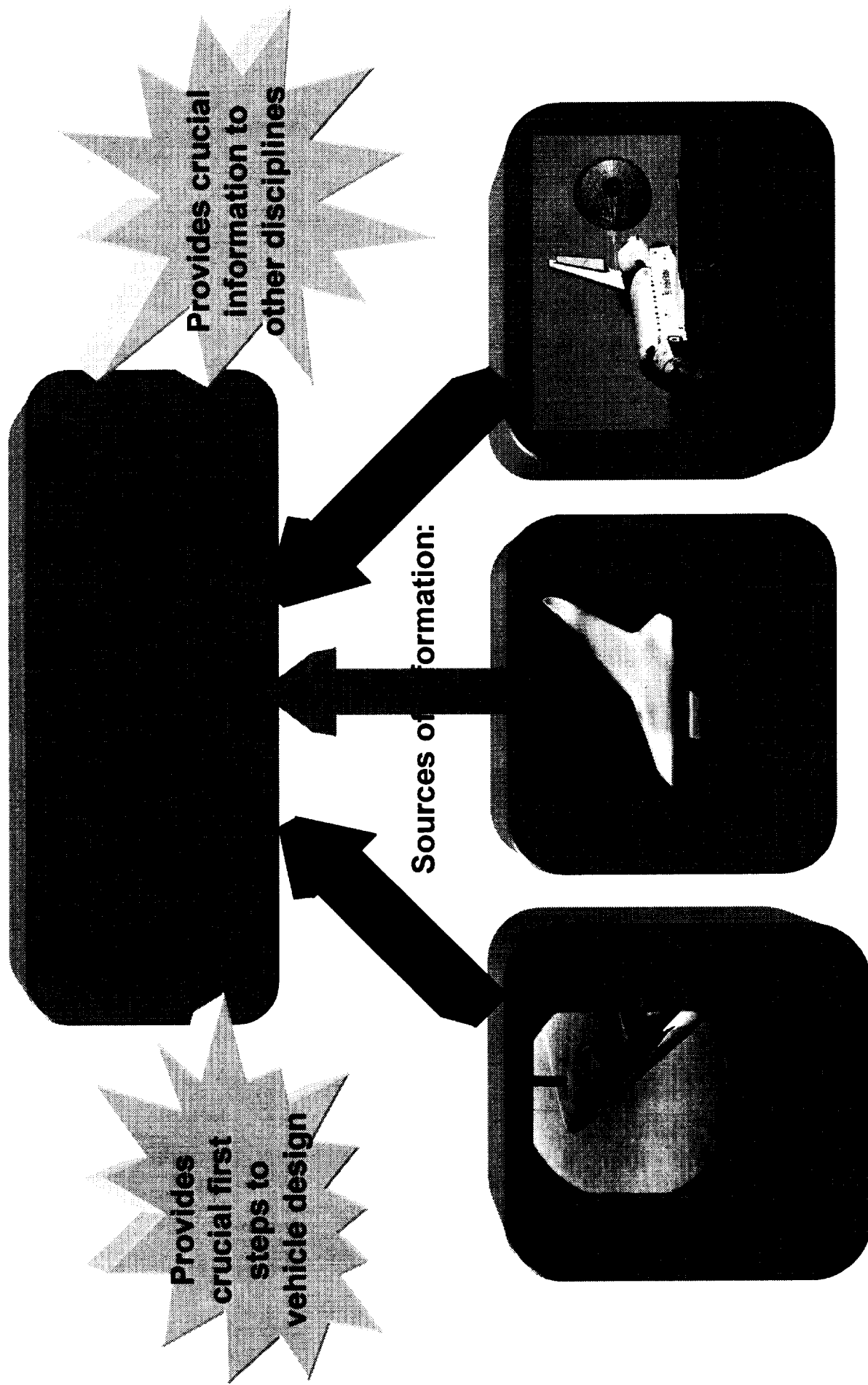
Airframe/TPS

INTEGRATED TPS/TPSS/TANK SYSTEM DEFINITION

- **Hierarchy of sizing methods are needed to support trade studies and concept assessment. Even low fidelity sizing must capture effect of major design drivers (e.g. geometry, size, load, ...)**
- **Major deficiencies exist in the types of material data needed to optimize integrated airframe (TPS/TPS Support/Tank systems) for both metallic and polymeric composite systems**
- **Critical interactions exist within airframe systems, and between airframe and vehicle systems. New analytical formulations and/or tools are needed to take advantage of those interactions which significantly improve or enable vehicle/system viability.**

Airframe/TPS

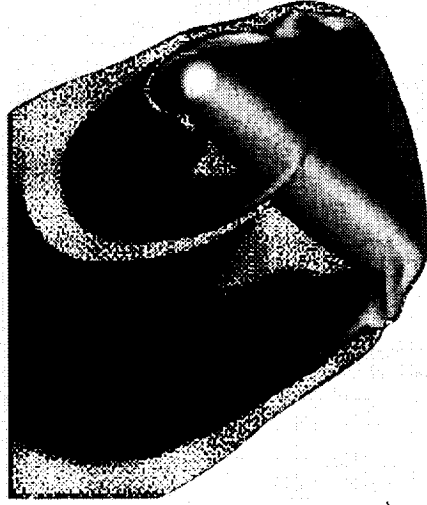
CONCLUDING REMARKS



Airframe/TPS - Aerothermodynamics

Aerothermodynamics

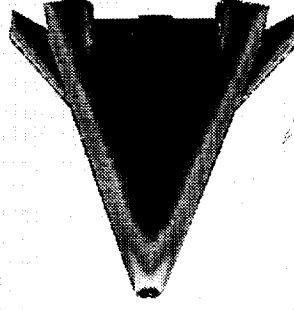
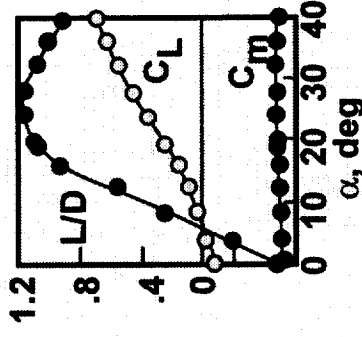
**Computational Fluid
Dynamics (CFD)**



**Calibrated
against
one
another
Applied
to flight
(ascent/
descent)**

**Ground-based
testing**

LMSW 603 B1001 at Mach 20



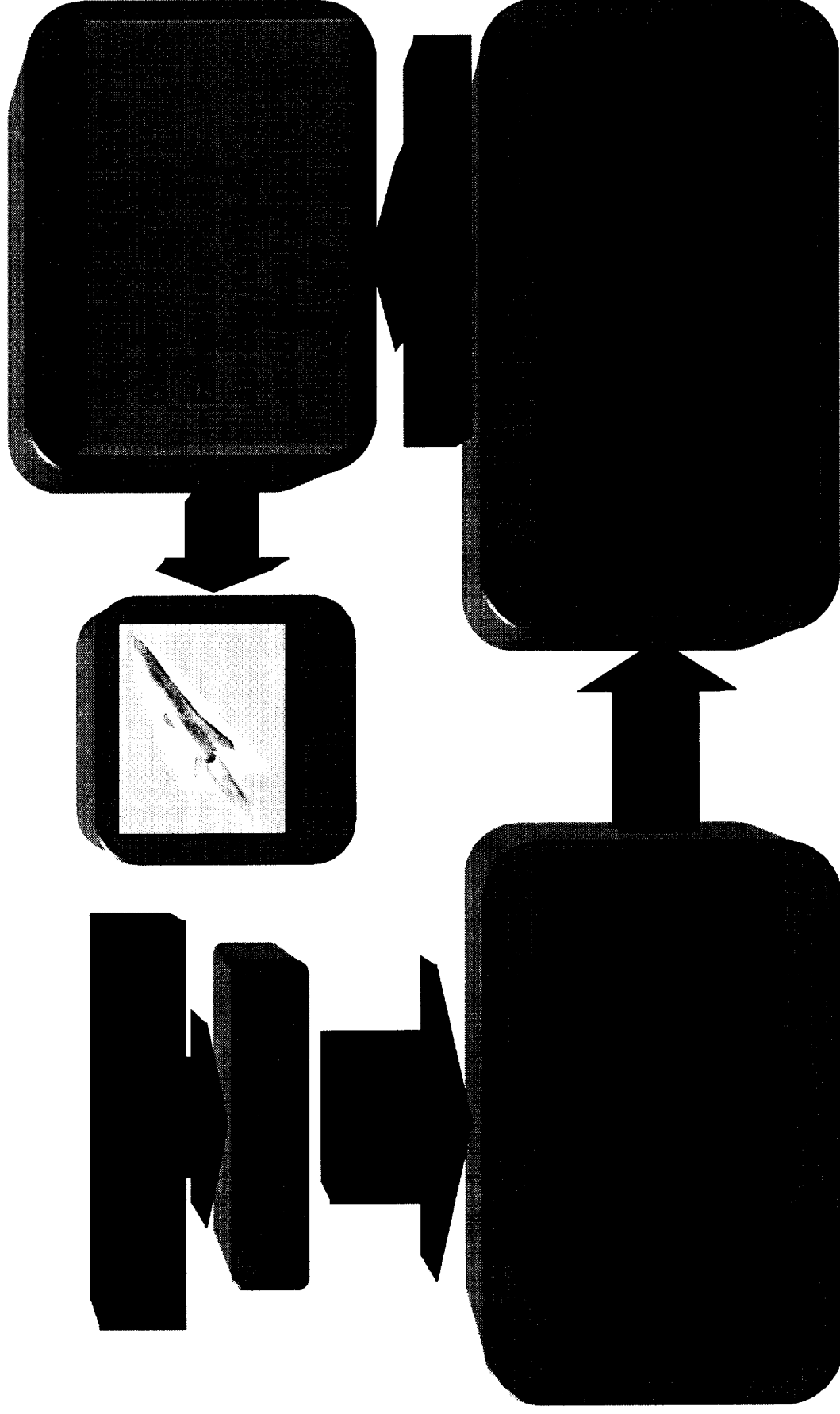
**Guidance,
Control
and
Navigation**

**Thermal Protection
System**

**Structural
Design**

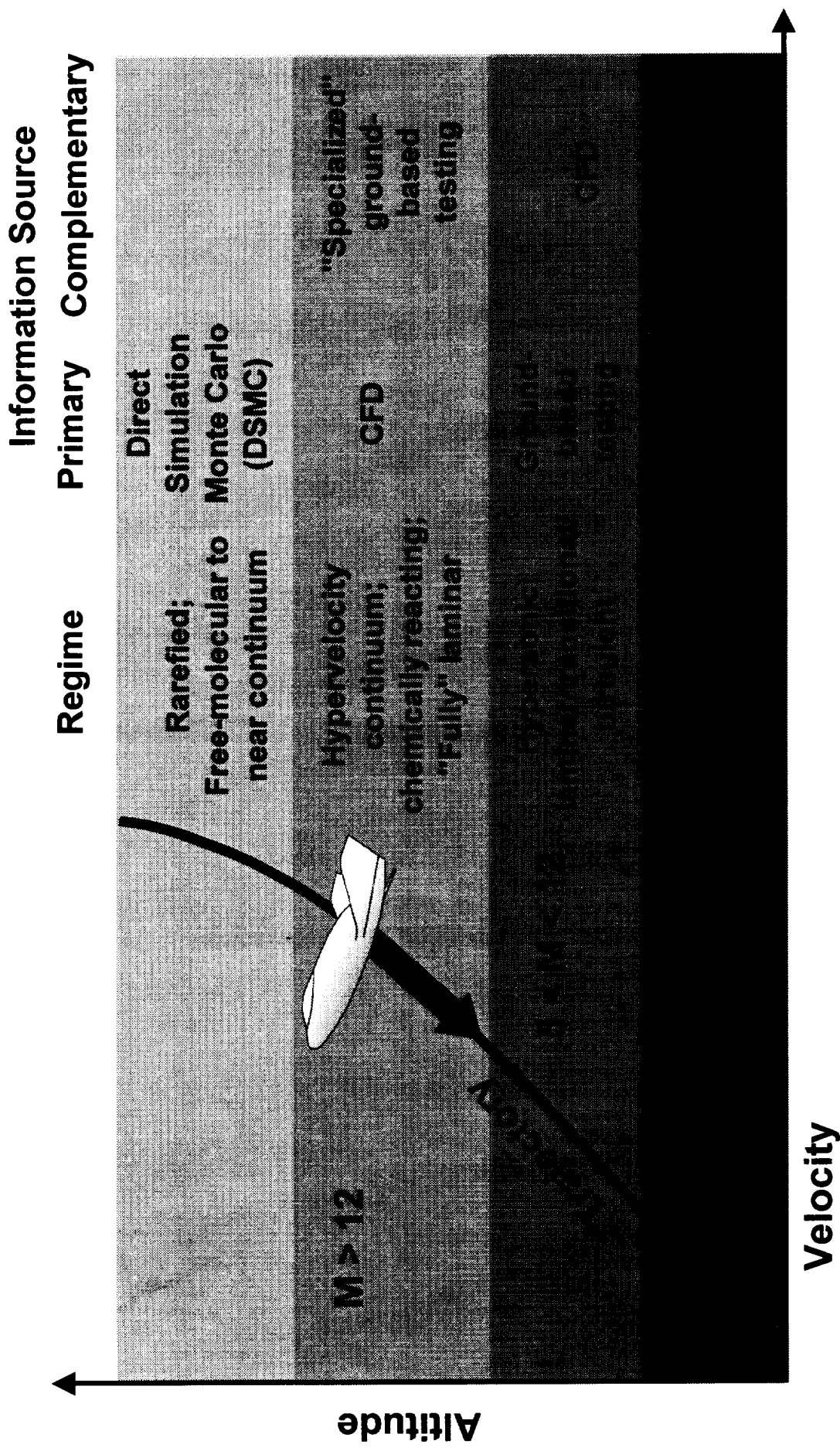
Materials

Airframe/TPS - Aerothermodynamics
**Aerothermodynamics Provides
Critical - Path Information**



Airframe/TPS - Aerothermodynamics

Aerothermodynamic Process



Airframe/TPS - Aerothermodynamics

Aerothermodynamic Methodology

Ames
Research Center
(thermal protection
systems)

Dryden Flight
Research Center

Johnson Space Center
(Crewed Aerospace Vehicles)

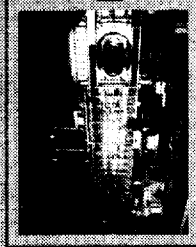
Marshall Space Flight
Center (Ascent)



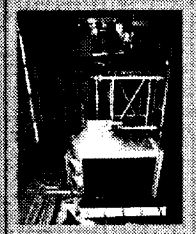
Langley Aerothermodynamic Facilities Complex (AFC)



20-Inch
Mach 6
Air



31-Inch
Mach 10
Air

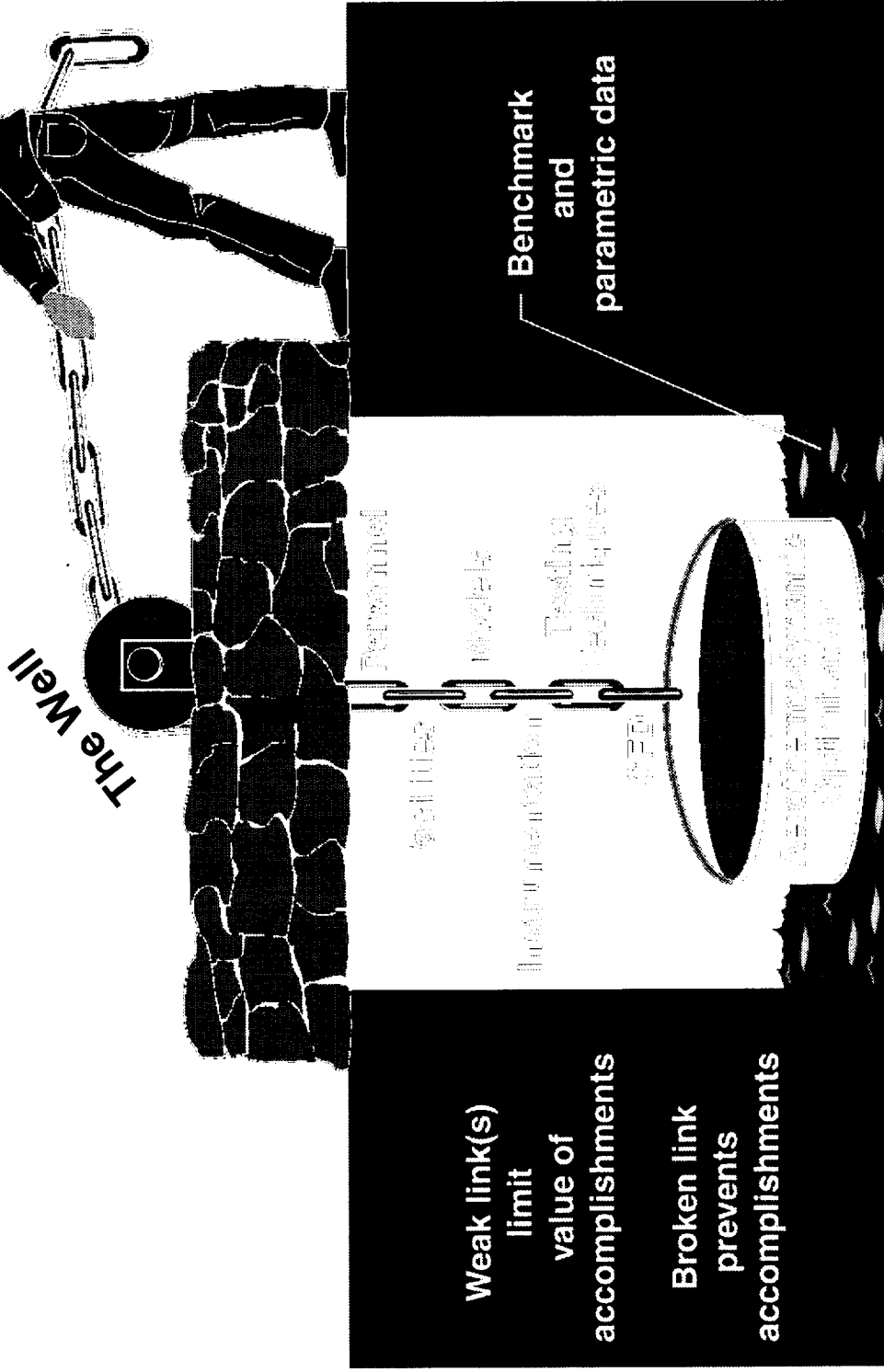


20-Inch
Mach 13-18
Real Gas
Simulation

Airframe/TPS - Aerothermodynamics

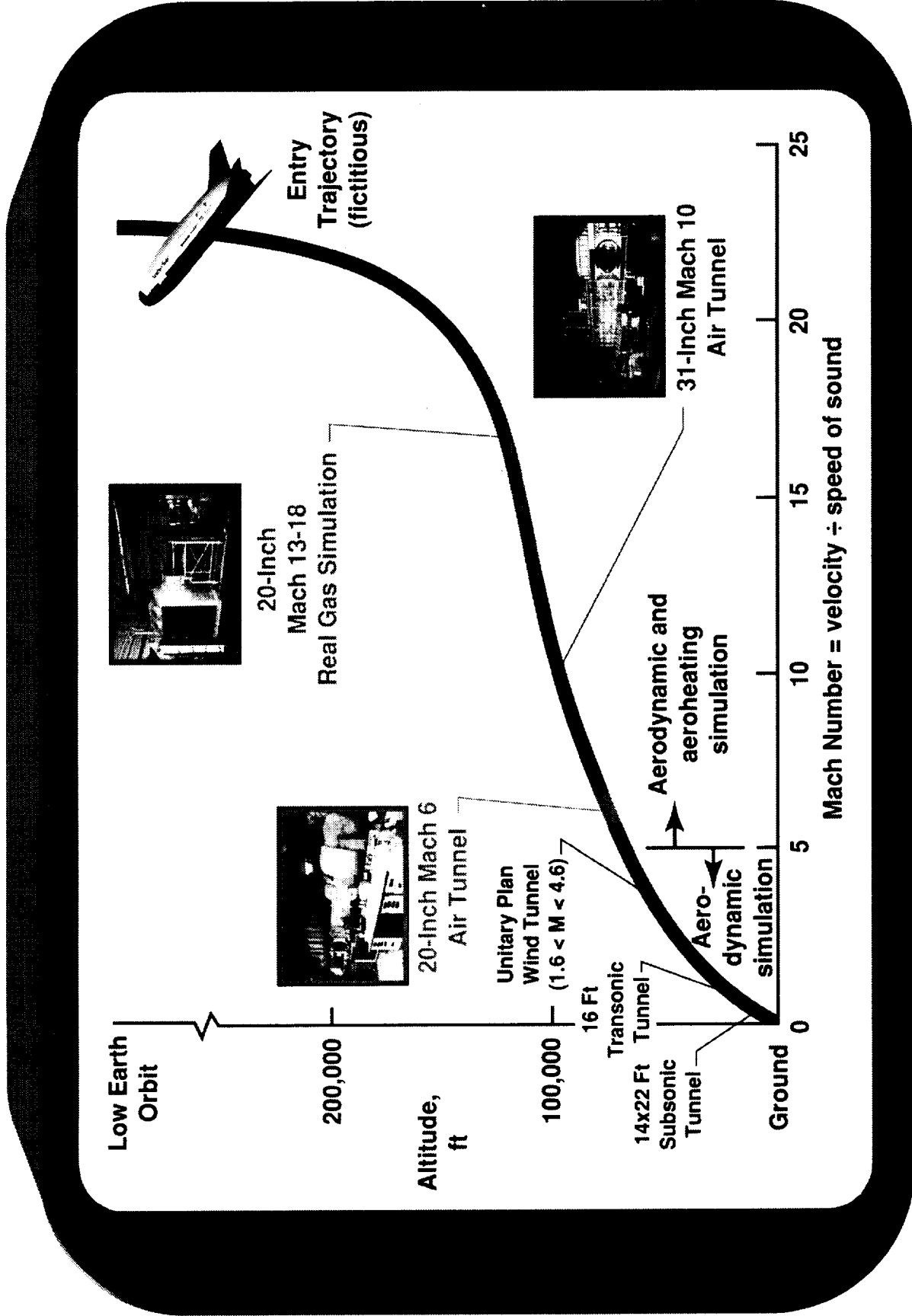
LaRC is NASA Lead For Aerothermodynamics

Aerothermodynamic customer



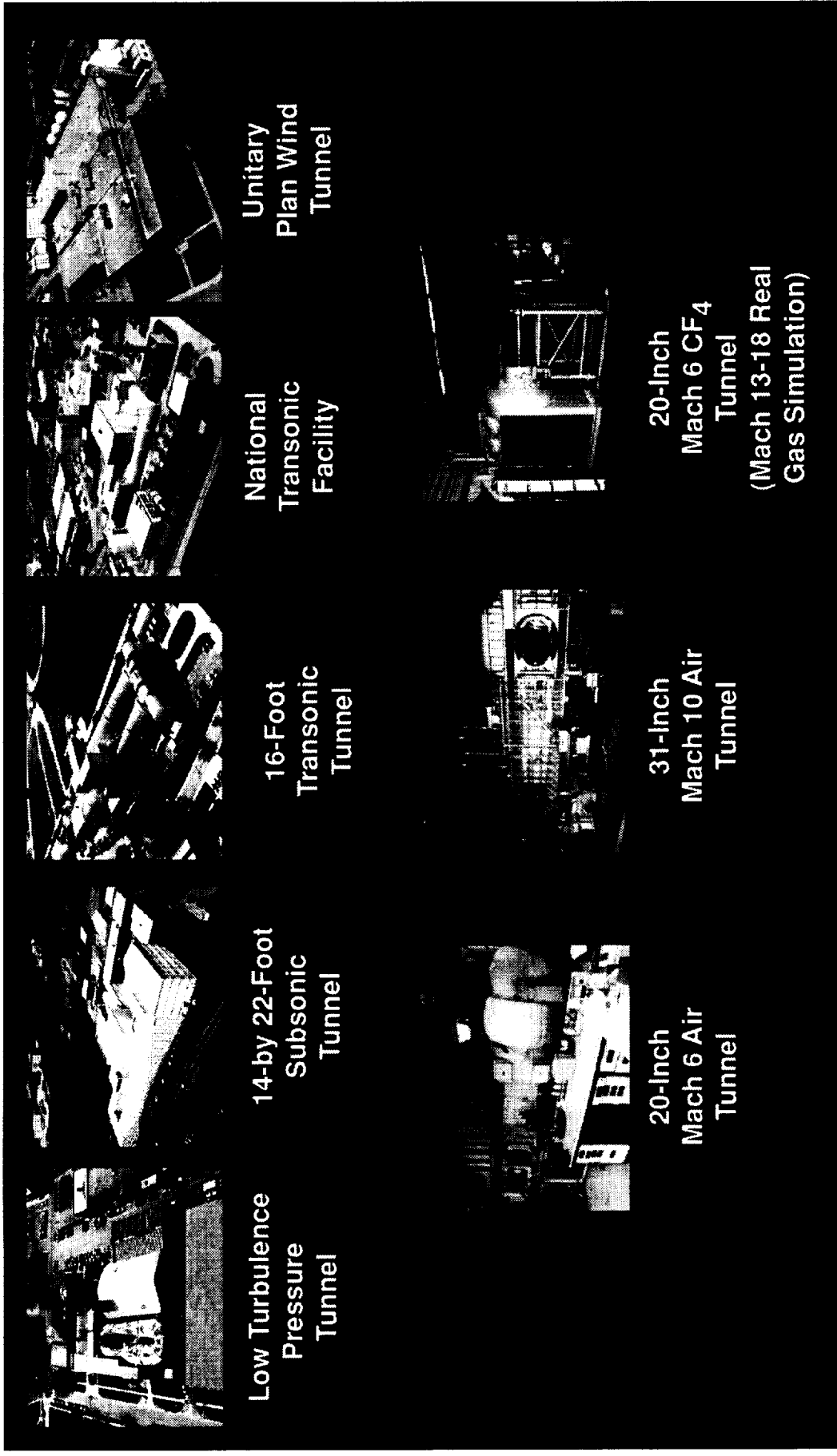
Airframe/TPS - Aerothermodynamics

Aerothermodynamic “Chain”



Airframe/TPS - Aerothermodynamics

Aerothermodynamic Flight Simulation Capability



Low Turbulence
Pressure
Tunnel

14-by 22-Foot
Subsonic
Tunnel

16-Foot
Transonic
Tunnel

National
Transonic
Facility

Unitary
Plan Wind
Tunnel

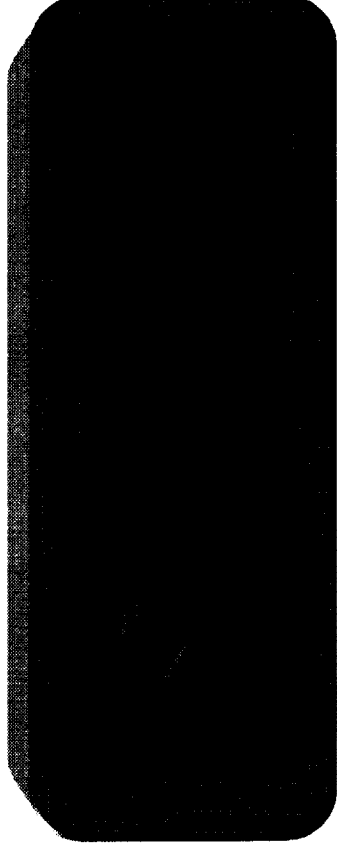
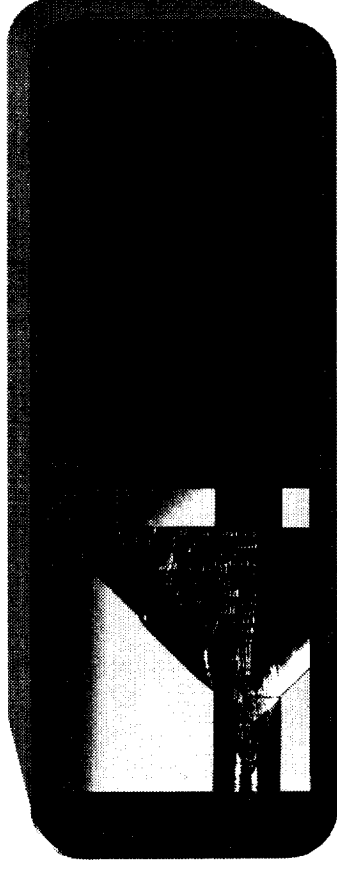
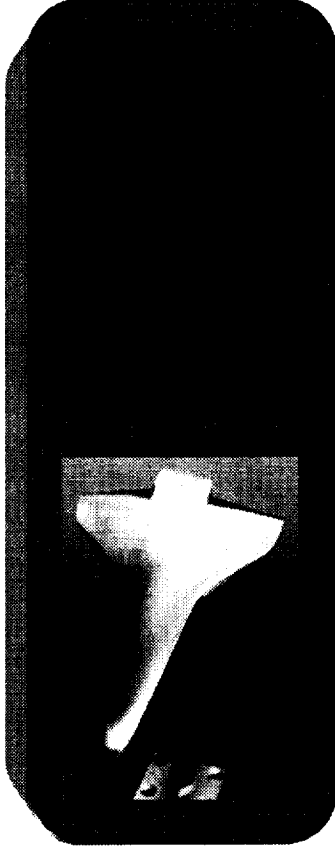
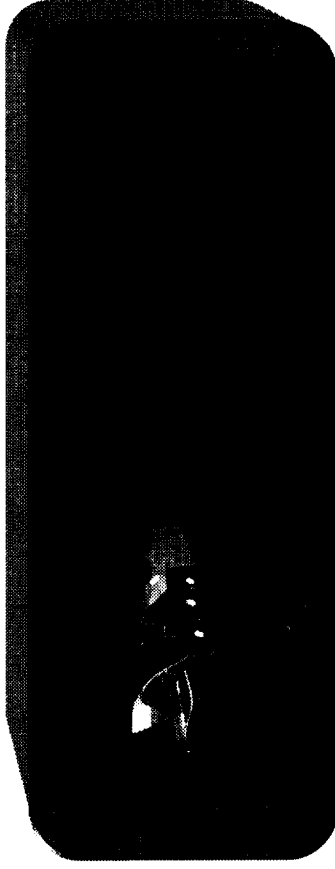
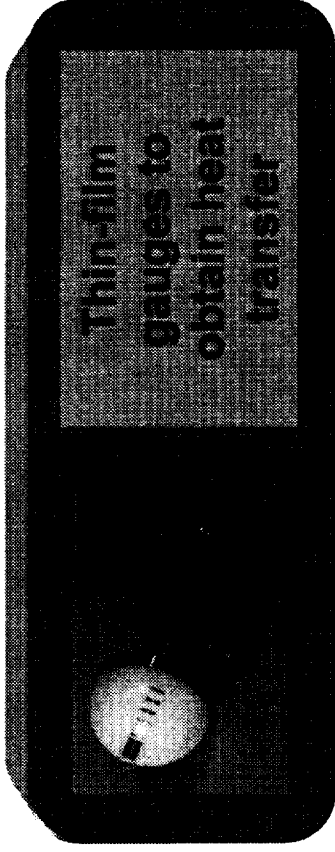
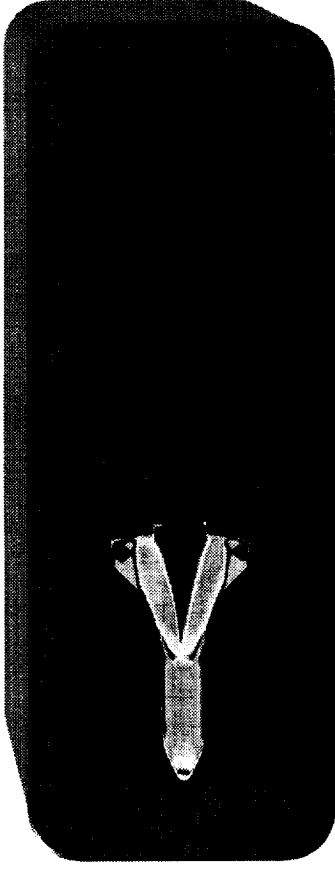
20-Inch
Mach 6 Air
Tunnel

31-Inch
Mach 10 Air
Tunnel

20-Inch
Mach 6 CF_4
Tunnel
(Mach 13-18 Real
Gas Simulation)

Airframe/TPS - Aerothermodynamics

LaRC Subsonic-to-Hypersonic Wind Tunnels



Airframe/TPS - Aerothermodynamics

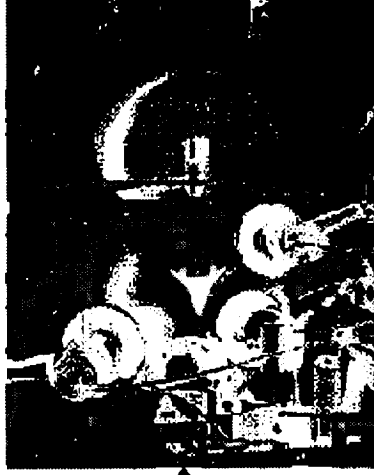
Testing Techniques

Vehicle Concept



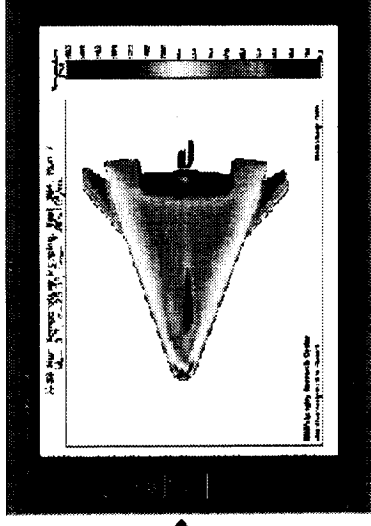
Model Fabrication

- Casting of ceramic models
- Rapid turnaround
- Complex shapes



Wind Tunnel Testing

- Two-color fluorescence
- State-of-art computerized acquisition system



Analysis of Measurements

- Nonlinear theory to infer accurate temperatures
- User-friendly computer program (IHEAT)

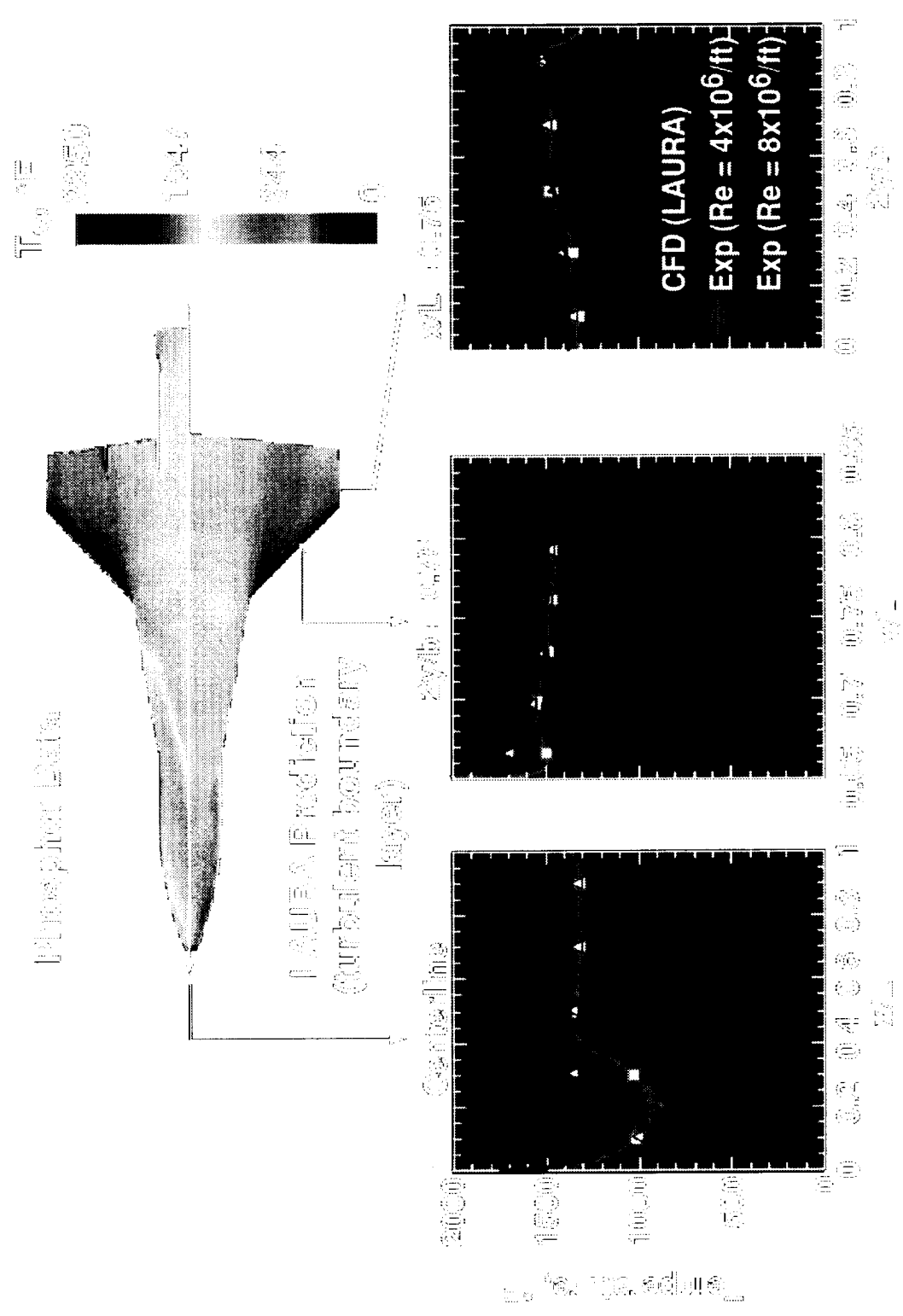


Aeroheating data to customers

Airframe/TPS - Aerothermodynamics

Phosphor Thermography Process

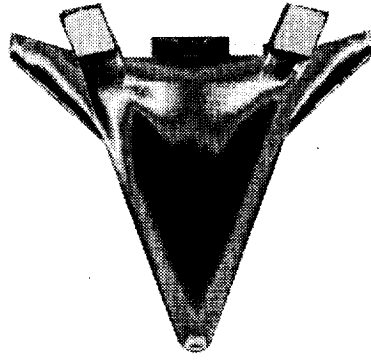
$M_\infty = 10.3$ $\alpha = 25^\circ$ $M_\infty = 10.3$ $\alpha = 0^\circ$



Airframe/TPS - Aerothermodynamics

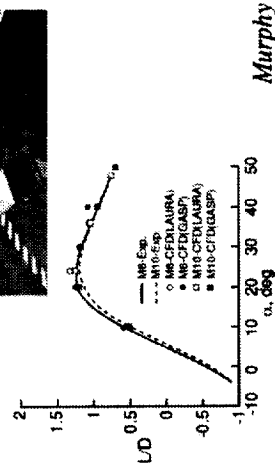
Extrapolation of Measurements to Flight

Surface Heating



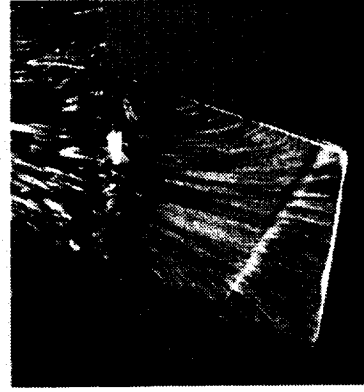
Horvath

Forces and Moments



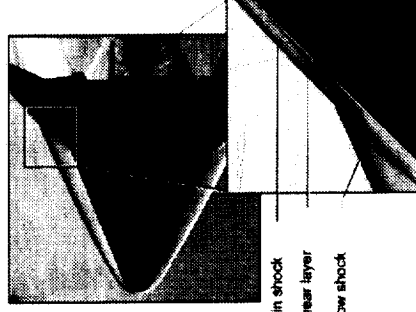
Murphy

Surface Streamlines



Horvath

Shock Waves



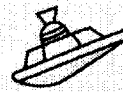
Horvath

Airframe/TPS - Aerothermodynamics

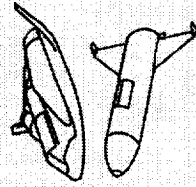
Complementary Measurements: X-33



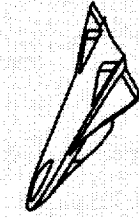
Space
Station



Aeroassist
Orbital Transfer



Space Transportation



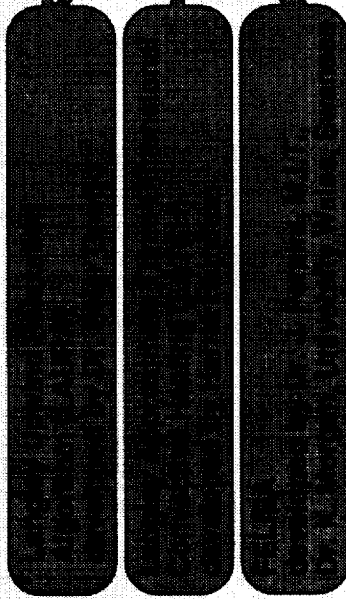
Aerospace
Planes

Free Molecular
Flow

Transitional
Flow

Continuum
Flow

LaRC Workhorse Codes for X-33/X-34/X-37



Navier Stokes*

Viscous Shock Layer

Inviscid/Boundary Layer

Direct Simulation Monte Carlo

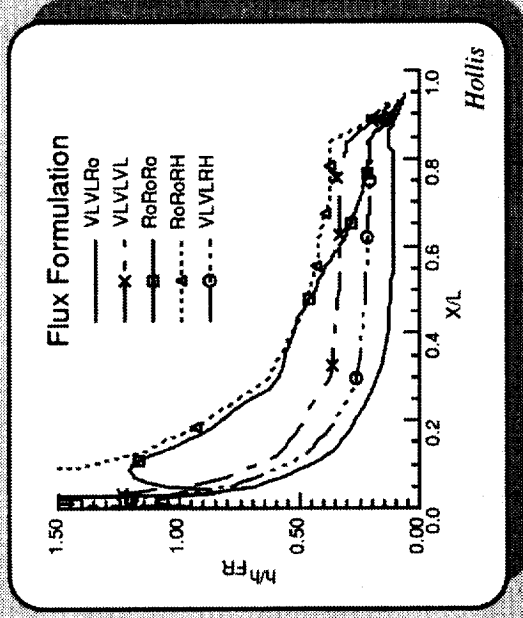
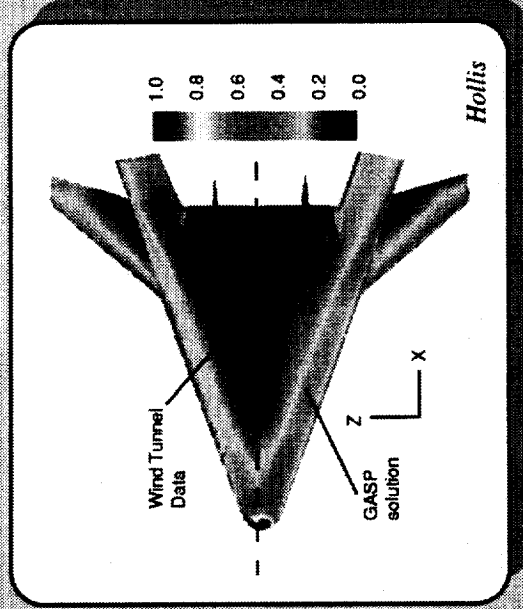
DSMC Analysis Code (DAC); 3D capability developed by Jay LeBeau (JSC) and Dr. Richard Wilmoth; physical modeling per Dr. Graeme Bird, Univ. of Sydney

* Also, GASP, registered trademark of AeroSoft, Inc.

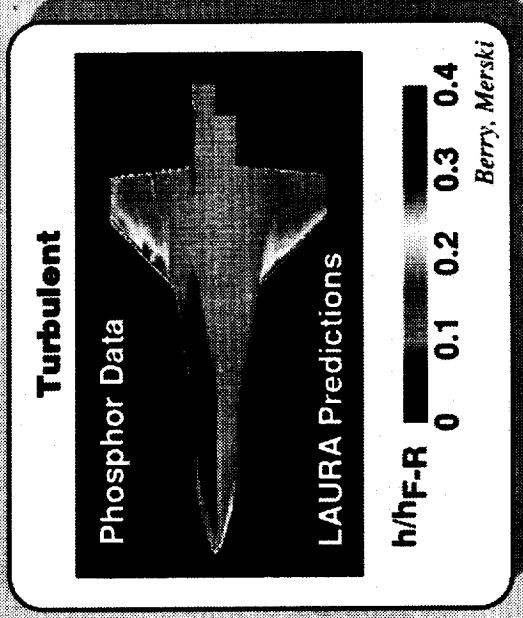
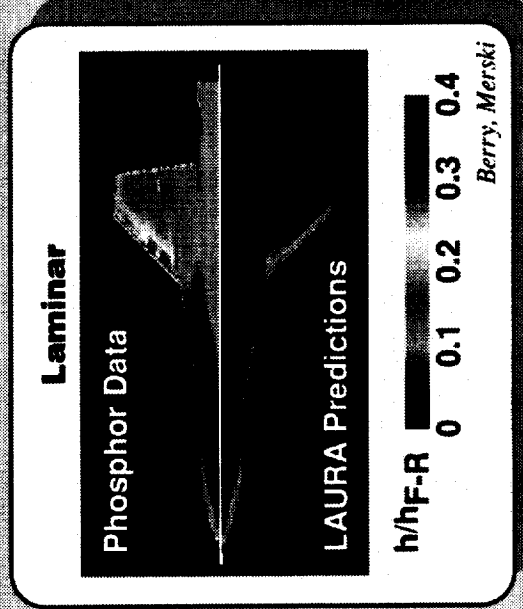
Airframe/TPS - Aerothermodynamics

Computational Fluid Dynamics (CFD)

X-33

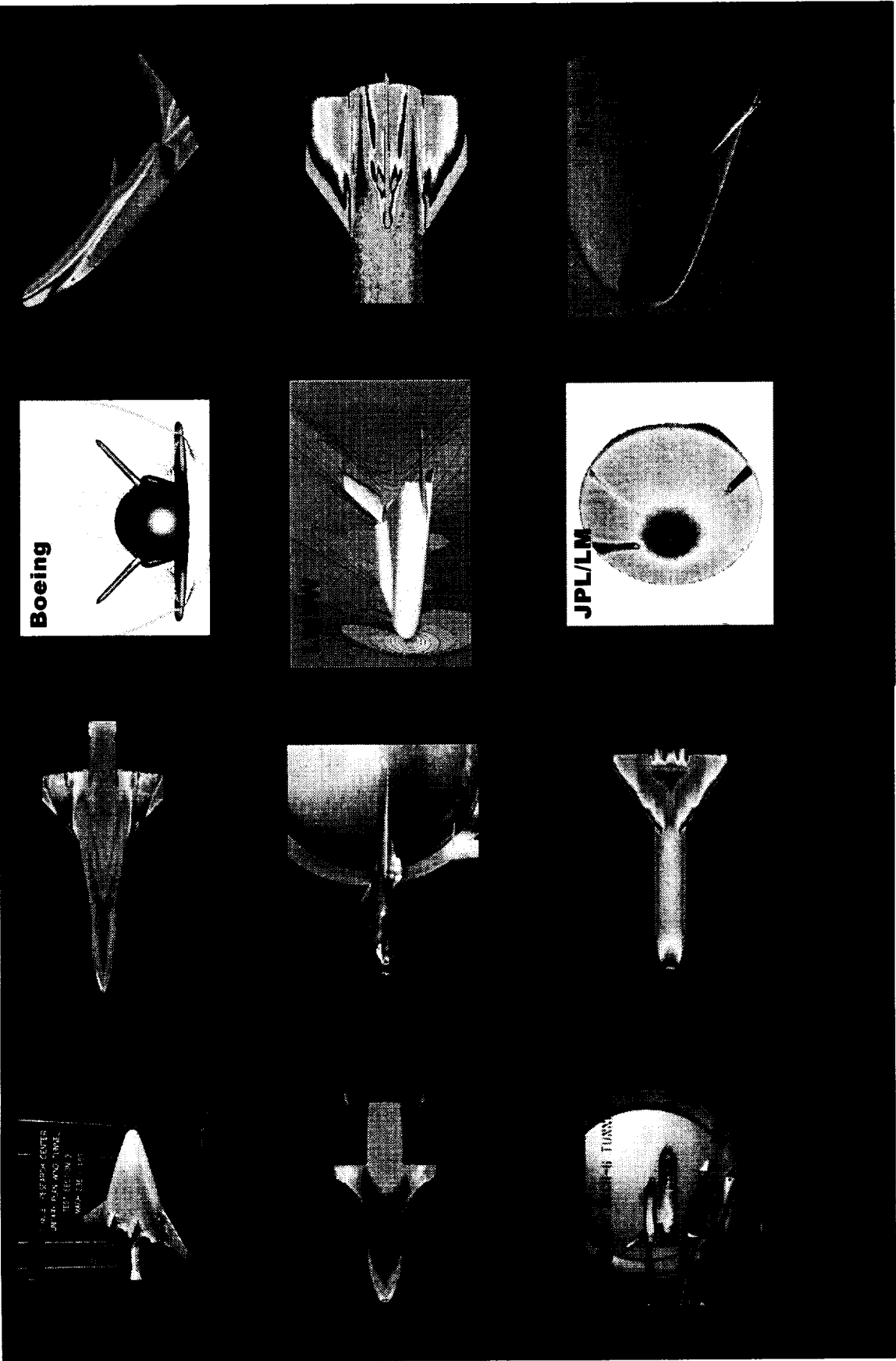


X-34



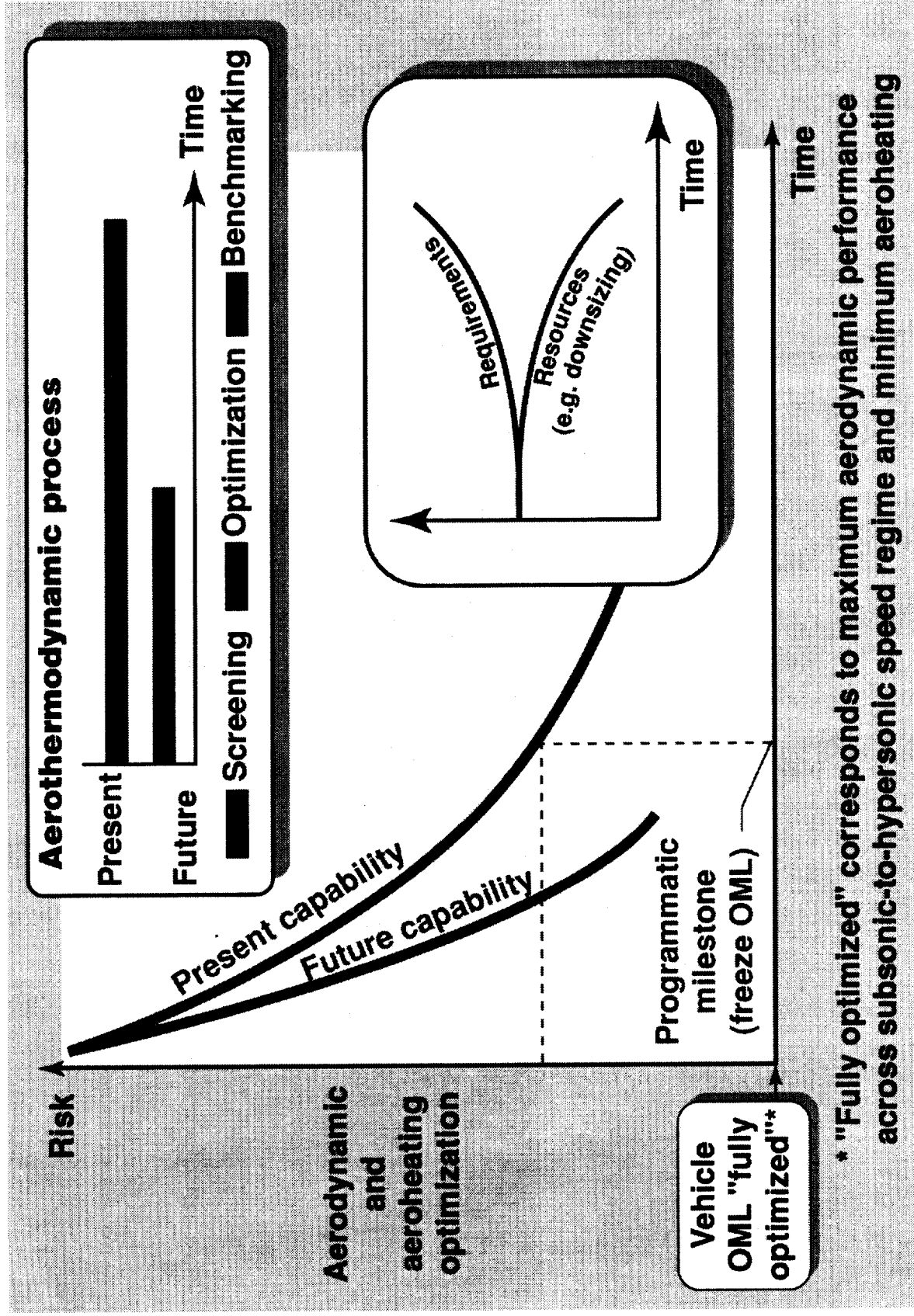
Airframe/TPS - Aerothermodynamics

Computational - Experimental Synergism



Airframe/TPS - Aerothermodynamics

Recent LaRC Aerothermodynamic Contributions



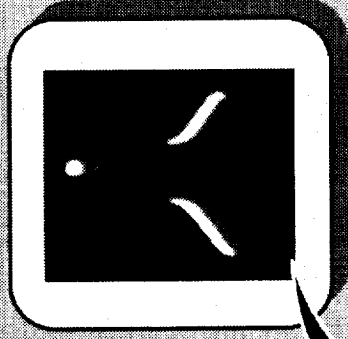
Airframe/TPS - Aerothermodynamics

Aerospace Vehicle Design: Risk vs. Time

Calibration/validation of experimental and computational tools via comparisons to flight data

- Aerodynamic performance/aeroheating characteristics extracted from flights

**BMDO/ISTEF
IR Image of
STS-96**



**LaRC Mapping
Code**

**Demonstrate on Orbiter
prior to X-33 application**



On-board and off-board aeroheating measurements

Airframe/TPS - Aerothermodynamics

Future Plans

- ♦ “Review of X-33 Hypersonic Aerodynamic and Aerothermodynamic Development”; ICA-0323

Richard A. Thompson, NASA Langley Research Center
Presented at 22nd International Congress of

Aeronautical Sciences, August 27 - Sept 1, 2000,
Harrogate, United Kingdom

Paper available:

<http://techreports.larc.nasa.gov/ltrs/PDF/2000/mtg/NASA-2000-22cicas-rat.pdf>

Extensive list of references

- ♦ Journal of Spacecraft and Rockets; Vol. 36, No. 2, Mar-Apr 1999
Special Section: X-34; pages 153-239
(collection of nine papers)

Airframe/TPS - Aerothermodynamics

Reference Sources for Recent RLV Studies

- | | |
|---|------------|
| ◆ 12:45 - 1:00 Introduction 2nd Gen RLV Airframe | S. Welch |
| ◆ 1:00 - 1:20 Airframe Design and Integration | S. Scotti |
| ◆ 1:20 - 1:40 Aerothermodynamics | C. Miller |
| ◆ 1:40 - 2:00 Structures and Materials | T. Johnson |
| ◆ 2:00 - 2:20 Tanks | D. Smith |
| ◆ 2:20 - 2:40 Thermal Protection Systems | M. Rezin |
| ◆ 2:40 - 3:00 Integrated Airframe Demonstrations | D. Glass |
| ◆ 3:00 - 3:05 BREAK | |
| ◆ 3:05 - 3:30 Introduction 3rd Gen RLV Airframe | D. Bowles |
| ◆ 3:30 - 3:55 Integrated Design and Analysis | T. Gates |
| ◆ 3:55 - 4:20 Integrated Thermal Str. & Materials | B. Jensen |
| ◆ 4:20 - 4:45 Thermal Protection Systems | S. Johnson |

2nd Gen Airframe/TPS - Structures and Materials:

Agenda